

## **TECHNICAL REPORT AMR-SS-07-04**

# CHNO ENERGETIC POLYMER SPECIFIC HEAT PREDICTION FROM THE PROPOSED NOMINAL/GENERIC (N/G) C<sub>P</sub> CONCEPT

James P. Billingsley

System Simulation and Development Directorate Aviation and Missile Research, Development, and Engineering Center

February 2007

Approved for public release. Distribution is unlimited.







#### **DESTRUCTION NOTICE**

FOR CLASSIFIED DOCUMENTS, FOLLOW THE PROCEDURES IN DoD 5200.22-M, INDUSTRIAL SECURITY MANUAL, SECTION II-19 OR DoD 5200.1-R, INFORMATION SECURITY PROGRAM REGULATION, CHAPTER IX. FOR UNCLASSIFIED, LIMITED DOCUMENTS, DESTROY BY ANY METHOD THAT WILL PREVENT DISCLOSURE OF CONTENTS OR RECONSTRUCTION OF THE DOCUMENT.

#### DISCLAIMER

THE FINDINGS IN THIS REPORT ARE NOT TO BE CONSTRUED AS AN OFFICIAL DEPARTMENT OF THE ARMY POSITION UNLESS SO DESIGNATED BY OTHER AUTHORIZED DOCUMENTS.

#### TRADE NAMES

USE OF TRADE NAMES OR MANUFACTURERS IN THIS REPORT DOES NOT CONSTITUTE AN OFFICIAL ENDORSEMENT OR APPROVAL OF THE USE OF SUCH COMMERCIAL HARDWARE OR SOFTWARE.

REPORT DOCUMENTATION PAGE  Form Appro OMB No. 07					
Public reporting burden for this collection of i	viewing instructions,	searching existing data sources,			
of information, including suggestions for redu	nd completing and reviewing this collection of in ucing this burden to Washington Headquarters S	Services, Directorate for Informati	ion Operations and Re	ports, 1215 Jefferson Davis Highway,	
1.AGENCY USE ONLY	the Office of Management and Budget, Paperwo  2. REPORT DATE	3. REPORT TYPE AND			
	February 2007	Final			
4. TITLE AND SUBTITLE			5. FUNDING N	UMBERS	
CHNO Energetic Polymer	Specific Heat Prediction Fro	m The Proposed			
Nominal/Generic (N/G) C <sub>P</sub>	Concept	•			
` / -	•				
6. AUTHOR(S)					
James P. Billingsley					
7. PERFORMING ORGANIZATION I	NAME(S) AND ADDRESS(ES)		8. PERFORMIN	G ORGANIZATION	
	esearch, Development, and		REPORT NU	MBER	
Engineering Command	, 1		TR-AMR	-SS-07-04	
ATTN: AMSRD-AMR-SS	-EG				
Redstone Arsenal, AL 3589	98-5000				
, and the second	GENCY NAME(S) AND ADDRESS(E	S)	10. SPONSORI	NG / MONITORING	
	(0,7,11,2,7,2,2,1,2,0)	-,		EPORT NUMBER	
44 OUDDI EMENTADY NOTES					
11. SUPPLEMENTARY NOTES					
12a. DISTRIBUTION / AVAILABILIT	Y STATEMENT			12b. DISTRIBUTION CODE	
Approved for public release, of					
,				A	
13. ABSTRACT (Maximum 200 Wo	rds)				
	specific heat (C <sub>P</sub> ) of CHNO	energetic polymers	relative to the	eir impact shock	
I	ented in MICOM TR-RD-SS	· ·		*	
_	on-report, RDECOM TR-A				
	etic materials. The motivation				
	whose $C_P$ was unknown. Thi				
	P for a relatively new CHNO			ar appreciation of the 14 S	
		••••••••••••••••••••••••••••••••••••••			
44 0110 1507 750110				45 NUMBER OF 54 050	
14. SUBJECT TERMS  CHNO Energetic Polymers	Specific Heat Heat Sensiti	vity Impact Shock		15. NUMBER OF PAGES 44	
CHNO Energetic Polymers, Specific Heat, Heat Sensitivity, Impact Shock Sensitivity, Nominal/Generic (N/G) C <sub>P</sub> , FOX-7					
				16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT	18. SECURITY CLASSIFICATION OF THIS PAGE	19. SECURITY CLASSIF OF ABSTRACT	FICATION	20. LIMITATION OF ABSTRACT	
UNCLASSIFIED	UNCLASSIFIED	UNCLASSII	FIED	SAR	

NSN 7540-01-280-5500

Standard Form 298 (Rev. 2-89) Prescribed by ANSI Std. Z39-18 298-102

#### **ACKNOWLEDGEMENTS**

The author gratefully acknowledges the assistance of Davidson Technologies, Inc. personnel in preparing this report. Ms. Judy Smith typed the manuscript and Mr. Shane Strickland assisted in preparing the computer-drawn graphs.

# TABLE OF CONTENTS

		<u>Page</u>
I.	INTRODUCTION	. 1
II.	BACKGROUND INFORMATION	. 2
	A. The $\Delta$ (v.e.) <sub>TR</sub> Concept B. The N/G C <sub>P</sub> Concept	. 2
III.	EXAMPLE OF C <sub>P</sub> PREDICTION FOR FOX-7 VIA THE N/G C <sub>P</sub> CONCEPT	. 6
IV.	DISCUSSION	. 8
V.	RECOMMENDATIONS	. 8
	REFERENCES	. 15
	APPENDIX A: COMPUTATION OF m <sub>AV</sub> , d <sub>1av</sub> AND MW FOR FOX-7	A-1
	APPENDIX B: SPECIFIC HEAT (C <sub>P</sub> ) UNITS CONVERSION RELATIONS	B-1

# LIST OF ILLUSTRATIONS

<u>Figure</u>	<u>Title</u>	<u>Page</u>
1.	FOX-7 C <sub>P</sub> Prediction Comparison	13
2.	Comparison of the Proposed N/C C <sub>P</sub> and the Calculated C <sub>P</sub> for FOX-7 From Reference 8	13

# LIST OF TABLES

<u>Table</u>	<u>Title</u>	Page
1.	The Generic C <sub>P</sub> per Average Atom for CHNO Explosives	9
2.	FOX-7 C <sub>P</sub> Prediction Via the Proposed N/G C <sub>P</sub>	10
3.	FOX-7 Computed Specific Heat Results for FOX-7 From Reference 8, (Fig. 6)	11
4.	FOX-7 C <sub>P</sub> (T) Prediction Comparison	12

#### I. INTRODUCTION

References 1 through 6 have demonstrated that the one-dimensional (1-D) plane impact shock sensitivity of several important energetic polymeric chemical CHNO compounds can be correlated to how much heat energy,  $\Delta$  (v.e.)<sub>TR</sub>, they can absorb before some type of reaction (melt, phase transition, deflagration, or detonation) occurs. The amount of internal energy that can be soaked up is directly related to the material's specific heat ( $C_P$ ) magnitude and variation with temperature (T). See Section II.A. for more details about the  $\Delta$  (v.e.)<sub>TR</sub> concept that relates  $C_P$  heat absorption to impact shock sensitivity.

The  $C_P$  (T) of CHNO energetic polymers is a very important property from both heat and impact shock sensitivity view points. As such, a simple  $C_P$  (T) estimation/prediction procedure is also desirable and very important when experimental data may not be available.

One attempt at a  $C_P$  (T) estimation/prediction scheme is documented in Reference 7 and briefly described in Section II.B. of this report. This is the so called Nominal/Generic (N/G)  $C_P$  per average atom concept.

An N/G  $C_P$  was proposed based on the observation that the  $C_P$ 's per average atom for CHNO explosives at a given temperature did not deviate very much from each other. So in lieu of experimental  $C_P$  data for a CHNO explosive compound, the N/G  $C_P$  could be utilized to make a  $C_P$  prediction.

A successful application of this  $C_P$  estimation procedure is documented in Section III of this report for the FOX-7 CHNO explosive compound. This application is called "successful" because the predicted  $C_P$  agreed remarkably well with  $C_P$  results computed by a much more complex and sophisticated analysis.

#### II. BACKGROUND INFORMATION

#### A. The $\Delta$ (v.e.)<sub>TR</sub> Concept

Essentially, the area  $\Delta$  (v.e.)<sub>TR</sub> under the  $C_P$  versus Temperature (T) plots between temperature limits (Experimental Test Temperature ( $T_{EXP}$ ) and Reactive Temperature ( $T_R$ )) is a measure of how much atomic vibratory energy explosives can absorb before a reaction occurs. The reaction may be melting, phase change, decomposition, burning, or even detonation. Thus, to a good approximation, it could be expected that if  $\Delta$ (v.e)<sub>TR</sub> amount of energy is suddenly added via impact shock loading, then a reaction may occur.

This  $\Delta$  (v.e.)<sub>TR</sub> concept, that impact shock sensitivity or shock induced reactivity of energetic materials could be related to their specific heat ( $C_P$ ) variation with temperature, was demonstrated in References 1 and 2 for RDX, TETRYL, PETN, TNT, and TATB, which are basic secondary reactive compounds.

In References 3 and 4, the  $\Delta$  (v.e.)<sub>TR</sub> ideas were demonstrated for HMX and HNS which are also important basic secondary explosive compounds. The impact shock response of these seven compounds ranges from very insensitive to highly sensitive. Most of these seven basic energetic compounds have been the main ingredient of useful explosive mixtures.

One such mixture is the plastic bonded explosive designated as PBX-9502 that is 95 percent TATB and 5 percent KEL-F80. PBX-9502 has been rather extensively tested via 1-D shock loading at various temperatures, and its thermal characteristics have also been experimentally explored. Consequently, with this much information available,  $\Delta$  (v.e.)<sub>TR</sub> concept computations were made for PBX-9502. The exploratory comparative results for this important energetic material were highly affirmative and are documented in References 5 and 6.

General details of the exploratory computation and experimental data comparisons involved in a general  $\Delta$  (v.e.)<sub>TR</sub> assessment are contained in the following paragraphs.

The thermal atomic vibratory energy increment,  $\Delta$  (v.e.)<sub>TR</sub>, is related to macroscopic critical particle (or mass) velocities ( $U_{PCR}$ ) and impact shock pressures via Equations (1) through (6).

For certain explosives, a good estimate of the critical particle velocity,  $U_{\text{PCR}}$ , where a reaction (or detonation) occurs is:

$$U_{P_{CRI}} = \sqrt{\frac{\Delta(v.e.)_{T_R}}{m_{AV}}}$$
 (1)

In some circumstances, a better estimate of the critical particle velocity is:

$$U_{PCR2} = \sqrt{\frac{2\Delta(v.e.)_{TR}}{m_{AV}}} = \sqrt{2} \quad U_{PCR1}$$
 (2)

where:

When good comparisons can be made, it has been generally found that experimental detonation particle (mass) velocity threshold values (  $U_{\text{PTH}}$  ) will be somewhere between  $U_{\text{PCRI}}$  and  $U_{\text{PCR2}}$ .

Certain explosives, when heated to higher and higher temperatures, melt before they explode (RDX and TNT, for example). This melting will require that the heat of fusion ( $\Delta H_F$ ) be absorbed by the material at  $T=T_{MELT}$  conditions before the temperature will increase [9, 10]. Consequently, if  $T_{EXP}$  is less than  $T_{MELT}$ , then the total heat absorbed from  $T=T_{EXP}$  to  $T=T_{EXPL}=T_R$  is:

$$\Delta(v.e.)_{\text{Tr}} = \int_{\text{Texp}}^{\text{Tmelt}} C_{\text{P}} dT + \Delta H_{F} + \int_{\text{Tmelt}}^{\text{Texpl}} C_{\text{P}} dT. \tag{4}$$

Therefore, for solid energetic materials which melt prior to explosion, then  $\Delta$  (v.e.)<sub>TR</sub>, as defined by Equation (4) is employed in Equations (1) and (2) to compute  $U_{PCR1}$  and  $U_{PCR2}$ , respectively. Note that melting is just one example of a phase transformation which may require an enthalpy increment ( $\Delta H_T$ ) to be activated. For example, HMX can exist in different solid polymorphic forms. At a certain temperature,  $T_T$ , one form may change to another form if the heat energy of transformation ( $\Delta$   $H_T$ ) is supplied. Therefore,  $\Delta H_T$  should be added to Equations (3) and (4) if  $T_R$  is greater than  $T_T$ .

Note that  $\Delta$  (v.e.)<sub>TR</sub> as defined by Equations (3) and (4) is actually an enthalpy increment ( $\Delta$ H). However, it was shown via numerical examples in Appendix B of Reference 1 that, under the experimental  $C_P$  acquisition conditions, the pressure times volume terms were minute compared to the  $C_P$  integral,  $\int_{T_{EXP}}^{T_R} C_p dT$ . Thus,  $\int_{T_{EXP}}^{T_R} C_p dT$  is essentially all of the internal energy difference caused by thermal stimulation during standard tests at atmospheric pressure to determine the specific heat characteristics.

Once  $\Delta$  (v.e.)<sub>TR</sub> and  $U_{PCR}$  values are computed, the corresponding shock velocity (U<sub>SCR</sub>) is ascertained from experimental data for  $U_S$  as a function of the particle velocity,  $U_P$ . The experimental relationship is usually linear and written empirically as:

$$U_S = C_O + S U_P \tag{5}$$

When  $U_P = U_{PCR}$  and  $U_S = U_{SCR}$  are determined, the shock pressure is computed from the following well known relation:

$$P_{S} = \rho_{e} U_{S} U_{P}, \qquad (6)$$

Where  $\rho_0 = \text{Material density (grams/cm}^3)$ .

Then  $U_{PCR}$ ,  $U_{SCR}$ , and  $P_{SCR}$  may be compared to experimental shock-induced reaction threshold information to check the validity of the above  $\Delta$  (v.e.)<sub>TR</sub> theory to denote reactive conditions under impact shock stimuli. The numerical computations involved in a  $\Delta$  (v.e.)<sub>TR</sub> assessment are straightforward and may be performed with a hand-held calculator.

It must be emphasized that any possible effect of pressure on  $C_P$  is not taken into account in the present analysis. The basic idea is that if a quantity of thermal vibration energy,  $\Delta$  (v.e.)<sub>TR</sub>, under quiescent conditions is able to create a reaction, then the same amount of energy added by an impact shock ( $e_i$  or  $e_t$ ) should also cause some type of reaction. The shock-induced reaction may not be the same type as the temperature induced reaction, but will nevertheless, be a reaction of some kind. It may be less or more severe than the thermally induced reaction.

The  $C_P$  unit of calorie/(atom  $^{\circ}K$ ) was employed in plots of  $C_P$  versus T information, which are shown in this report. This is because the Boltzman constant,  $k_B$ =0.33 x  $10^{-23}$  calories/(atom  $^{\circ}K$ ) and the maximum  $C_P$  at high temperatures for many materials is 3  $k_B \approx 1.0$  x  $10^{-23}$  calories/(atom  $^{\circ}K$ ). This is a good mnemonic reference level for comparison purposes. It was noted in Reference 1 that the average  $C_P$  per atom for most polymers never reaches the 3  $k_B$  level before a reaction (phase change, melting, glass-to-rubber transition, or even detonation) occurs.

Actually,  $C_P$  for some atoms, or combinations of atoms, probably reaches the 3  $k_B$  level and causes a reaction at some  $T_R$ . But  $C_P$  for a large number of atoms remains much less than 2  $k_B$ . Thus, a large amount of the possible thermal vibratory energy is never activated and the average  $C_P$  per atom remains relatively low [14]. In many cases, important temperature

induced reactions occur near the average  $C_P \approx 2 k_B$  level at moderate temperatures (400 to 600 °K).

#### B. The N/G C<sub>P</sub> Concept

It was first documented in Reference 1 that the  $C_P$  [Cal/(atom  $^{\circ}$ K)] magnitudes (at a given T) for five solid explosive compounds (RDX, TETRYL, PETN, TNT, and TATB) did not differ very much from each other. In Reference 3, it was demonstrated that the  $C_P$  for HNS was very close to that for TATB, and the  $C_P$  for HMX was somewhat less than the TATB  $C_P$  at the higher temperatures. Liquid TNT has a larger  $C_P$  (at a given T) than these solid energetic compounds. These statements are corroborated by the experimental  $C_P$  information exhibited in Figures 1 through 4 of Reference 7.

So with two exceptions (melted TNT and  $\delta$  – HMX), the  $C_P$  per atom of five important secondary energetic compounds all had very similar magnitudes near that for TATB. HNS also has the same magnitude and trend (non-linear variation) with temperatures as TATB. The other explosives (RDX, TNT, TETRYL and PETN)  $C_P$  have a linear variation (a + b T) over most of their temperature range.

The amount of heat energy these compounds and mixtures can absorb varies considerately. TATB and PBX-9502 can soak up more heat energy,  $\Delta$  (v.e.)<sub>TR</sub>, than the others by a considerable margin. That is, their reactive temperatures,  $T_R$ , were much greater than the other compounds (more than 100 °K for TNT, HNS and HMX and over 200 °K for TETRYL, PETN, and RDX).

So based on the above remarks and similar remarks in References 1, 3, and 5, a nominal  $C_P$  per average atom was proposed and defined as shown in Figure 2 and listed in Table 1. These nominal  $C_P$  magnitudes at Room Temperature (RT) and above are very similar to those of TATB, particularly at the high temperatures. Below RT, the proposed N/G  $C_P$  magnitudes are very close to (or equal to) the RDX  $C_P$  values and TNT  $C_P$  values near absolute zero.

In Reference 7, two examples (TNT and HMX) were selected for comparative computations where  $U_{PCR}$  and  $P_{SCR}$  were calculated via the experimental  $C_P$  and the N/G  $C_P$ . TNT has a high  $C_P$  relative to the N/G  $C_P$  and HMX has a low  $C_P$  relative to the N/G  $C_P$ . Two cases were considered for TNT where the large liquid  $C_P$  was included in Case 1.

Consequently, excluding Case 1 for TNT where the large liquid  $C_P$  had to be included in the  $U_{PCR}$  (EXP) and  $P_{SCR}$  (EXP) computations, then for TNT (Case 2), PBX-9502, and HMX, the maximum percentage differences between the (EXP)  $C_P$  and (N/G)  $C_P$  results were bounded by:

$$\frac{\Delta U_{\text{Pcr}}}{U_{\text{Pcr}}(EXP)} \ = \ \frac{U_{\text{Pcr}}(N/G) - U_{\text{Pcr}}(EXP)}{U_{\text{Pcr}}(EXP)} \ X \ 100.0 < | \ 2.5\% \ | \eqno(9)$$

$$\frac{\Delta P_{SCR}}{P_{SCR}(EXP)} = \frac{P_{SCR}(N/G) - P_{SCR}(EXP)}{P_{SCR}(EXP)} \times 100.0 < |3.5\%|$$
(10)

These small differences provide considerable credibility for the proposed N/G C<sub>P</sub> per average atom concept for most CHNO energetic materials.

#### III. EXAMPLE OF C<sub>P</sub> PREDICTION FOR FOX-7 VIA THE N/G CONCEPT

The relatively new CHNO explosive compound, FOX-7 (1, 1-Diamino-2, 2-Dinitroethylene), has generated a considerable amount of interest among explosive investigators within the last decade. References 8 through 11 are a good sampling of the experimental and theoretical efforts devoted to this energetic material.

However, there has been no experimental determination of FOX-7's specific heat, but there is a rather sophisticated and complex theoretical prediction of  $C_P$  as a function of T given in Reference 8. As such, this fits our search criteria for a good CHNO energetic compound to apply the N/G  $C_P$  concept for predicting the  $C_P$  magnitudes.

Thus, FOX-7 is a good choice for a C<sub>P</sub> prediction example Via the N/G C<sub>P</sub> concept because:

- 1. No experimental  $C_P$  (T) data for FOX -7 were available and, consequently, could not have influenced the proposed N/G  $C_P$  magnitudes.
- 2. There was, however, a theoretical computation of  $C_P$  (T) for FOX-7 that was available for comparison with  $C_P$  (T) from any other predictive scheme or with experimental data.

In Appendix A, the Molecular Weight (MW) and  $m_{AV}$  (mass of an <u>average</u> atom) for FOX-7 are computed. These values are:

$$m_{AV} = 1.756239 (10^{-23}) \text{ grams/atom}$$

Also for 
$$C_2 H_4 N_4 O_4 = C_i H_j N_k O_1$$

$$q = i + j + k + 1 = 14.0$$
 atoms/molecule

From various sources, Avogadro's number is:

$$N_{AV} = 6.02252 (10^{+23}) = molecules/MW$$

= Number of molecules per mole (MW) in a chemical compound. Also  $N_{AV}$  is the number of atoms in an Atomic Weight (AW) of an element.

As shown in Appendix A, the number of atoms per mole (MW) of FOX-7 can be computed by two different ways. These are:

1. NAPMW<sub>1</sub> = 
$$\frac{MW}{m_{AV}}$$
 =  $\frac{Grams/MW}{Grams/Atom}$  =  $\frac{148.086}{1.756329(10^{-23})}$   
= 84.315638 (10<sup>+23</sup>) Atoms/MW  
2. NAPMW<sub>2</sub> =  $q N_{AV}$  =  $\frac{Atoms}{Molecule} \cdot \frac{Molecules}{MW}$   
= (14.0) (6.02252 x 10<sup>+23</sup>)  
= 84.315280 (10<sup>+23</sup>)  $\frac{Atoms}{MW}$ 

These results for NAPMW agree to three decimal places and so

NAPMW = 
$$84.315 (10^{+23}) \text{ Atoms/MW}$$

is used in the following computations to predict the FOX-7  $C_P$  from the N/G  $C_P$ . That is, for a given temperature (T):

$$\begin{split} FOX\text{--}7 \ C_P &= \quad (NAPMW) \qquad C_{P_{N/G}} \\ &= \quad \left(\frac{Atom}{MW}\right) \left(\frac{Cal.}{Atom - {}^{\circ}K}\right) \qquad = \quad \left(\frac{Cal.}{MW - {}^{\circ}K}\right) \end{split}$$

Also:

$$FOX-7 \ C_P \quad = \quad \ 4.184 \ C_P \left( \frac{Cal.}{MW - {}^{\circ}K} \right) \qquad \quad = \quad \left( \frac{Joules}{MW - {}^{\circ}K} \right)$$

These  $C_P$  prediction computations are listed in Table 2 for a wide range of temperatures (100 to 650 °K). The results,  $C_P$  [Joules/ (MW-°K)], are plotted versus T (°K) in Figure 1.

Table 3 lists the FOX-7 theoretical  $C_P$  [Joules/ (MW- $^{\circ}$ K)] results from Reference 8, (Fig. 6). This information is also plotted in Figure 1 for comparative purposes. The  $C_P$  magnitudes from these two predictive schemes are in remarkably close agreement.

Table 3 also contains the conversion of the Reference 8 FOX-7  $C_P$  in [Cal/ (Atom-°K)]. These results are plotted in Figure 2 along with the proposed N/G  $C_P$  that is in the same units. As expected from the Figure 1 comparison, the FOX-7  $C_P$  and the N/G  $C_P$ , on a per average atom basis, agree exceptionally well.

#### IV. DISCUSSION

As shown in Figure 1, the FOX-7  $C_P$  (T) magnitudes, predicted from the proposed N/G  $C_P$  concept in the present report and the theoretical  $C_P$  results in Reference 8, compare exceptionally well. Table 4 lists the magnitude and the percentage differences. The percentage difference is within 6.0 percent for 100 °K  $\leq$  T  $\leq$  400 °K.

The largest negative difference (-5.9 percent) occurs at  $100\,^{\circ}$ K and the greatest positive difference (+3.9 percent) occurs at  $400\,^{\circ}$ K. So the predictive trends (or slopes) with temperature are slightly different but the  $C_P$  magnitudes are quite comparable. This is remarkable, considering the differences in sophistication and complexity between the predictive methods.

#### V. RECOMMENDATIONS

As a result of the present analysis and comparison of predicted  $C_P$  (T) results for the FOX-7 CHNO explosive compound, the following recommendations are made:

- 1. Experimental C<sub>P</sub> data for FOX-7 should be acquired:
  - a. To compare with both  $C_P$  predictions.
  - b. To have the authoritative  $C_P$  for use in the  $\Delta$  (v.e.)<sub>TR</sub> concept impact shock sensitivity computations.
- 2. The temperature range should be from 100 °K to a practical upper limit from experimental considerations. See Reference 11 for documentation of early work on thermal sensitivity.
- 3. Plane shock impact experiments should be performed to ascertain a lower threshold detonation limit for  $U_P$  or  $P_S$ . This should be done for more than one temperature ( $T_{EXP}$ ). Reference 5 documents a  $\Delta$  (v.e.)<sub>TR</sub> concept shock sensitivity analysis for PBX-9502 at four  $T_{EXP}$  temperatures (218, 300, 348 and 525 °K).
- 4. Predict the FOX-7 C<sub>P</sub> (T) variation using Satoh's scheme for submolar group additive contributions to the total C<sub>P</sub> of a polymer compound. See References 12 and 13 for further information on this additive property as applied to inert polymers.

With regard to recommendations 1, 2 and 3, this experimental data may be available for FOX-7, but if so, it is not known to the author of this report.

With regard to recommendation No. 4, some rather affirmative preliminary C<sub>P</sub> predictions via the Satoh submolar group additive property have been made for TNT and RDX. Similar computations will be made for FOX-7 and all these C<sub>P</sub> predictive results for energetic polymer compounds will be published in a forthcoming report.

The above recommendations/applications/comments concerning explosives in this report should also be valid for solid CHNO rocket propellants.

Table 1. The Generic  $C_P$  per Average Atom for CHNO Explosives

T	T	$C_{P}$	Remarks
°K	°C	<u>Cal</u> Atom - °K	~
0.0	-273	0.0000	<b>†</b>
7.0	-266	$0.001344(10^{-23})$	Same as
15.0	-258	$0.01463(10^{-23})$	RDX (S.C.)***
78.0	-195	$0.1629(10^{-23})$	<b>\</b>
100.0	-173	$0.2000(10^{-23})$	<b>†</b>
150.0	-123	$0.2611(10^{-23})$	
200.0	-73	$0.3222(10^{-23})$	
218.0	-55	$0.3442(10^{-23})$	
250.0	-23	$0.3833(10^{-23})$	See Note 1
293.0	20	$0.4359(10^{-23})$	
300.0	27	$0.4444(10^{-23})$	
348.0	75	$0.5031(10^{-23})$	
350.0	77	$0.5056(10^{-23})$	
400.0	127	$0.5667(10^{-23})$	
450.0	177	$0.6277(10^{-23})$	
500.0	227	$0.6889(10^{-23})$	
525.0	252	$0.7194(10^{-23})$	
550.0	277	$0.7500(10^{-23})$	<b>★</b>
600.0	327	$0.7600(10^{-23})$	1
650.0	377	$0.7600(10^{-23})$	Same as
669.0	396	$0.7600(10^{-23})$	TATB
700.0	427	$0.7600(10^{-23})$	<b>\</b>

\*\*\*S.C. = Single Crystal

Note 1: 
$$C_P \left[ \text{Cal/(Atom - }^{\circ} \text{K} \right] = \left[ 0.2000 + 0.001222 \left( \text{T} - 100 \right) \right] 10^{-23}$$
  
=  $\left[ 0.07780 + 0.001222 \, \text{T} \left( ^{\circ} \text{K} \right) \right] 10^{-23}$ 

For:  $100 \le T (^{\circ}K) \le 550$ 

Table 2. FOX-7  $C_P$  Prediction Via the Proposed N/G  $C_P$ 

Т	Т	C <sub>P</sub> (N/G)	NAPMW	C <sub>P</sub> FOX-7	C <sub>P</sub> FOX-7	C <sub>P</sub> FOX-7
°K	°C	Cal.	Atoms	Cal.	Joules	Joules
IX	C	Atom – °K	MW	MW - °K	MW - °K	Gram - °K
0	-273	0.0	84.315 (10 <sup>+23</sup> )	0.0	0.0	0.000
100	-173	$0.2000 (10^{-23})$	84.315 (10 <sup>+23</sup> )	16.863	70.555	0.476
150	-123	$0.2611 (10^{-23})$	$84.315 (10^{+23})$	22.015	92.109	0.622
200	-73	$0.3222(10^{-23})$	84.315 (10 <sup>+23</sup> )	27.166	113.664	0.768
250	-23	$0.3833(10^{-23})$	84.315 (10 <sup>+23</sup> )	32.318	135.218	0.913
300	27	$0.4444(10^{-23})$	84.315 (10 <sup>+23</sup> )	37.470	156.773	1.059
350	77	$0.5056 (10^{-23})$	$84.315 (10^{+23})$	42.630	178.363	1.204
400	127	$0.5667 (10^{-23})$	84.315 (10 <sup>+23</sup> )	47.781	199.917	1.350
450	177	$0.6277 (10^{-23})$	84.315 (10 <sup>+23</sup> )	52.925	221.436	1.495
500	227	$0.6889 (10^{-23})$	$84.315 (10^{+23})$	58.085	243.026	1.641
550	277	$0.7500 (10^{-23})$	84.315 (10 <sup>+23</sup> )	63.236	264.580	1.787
600	323	$0.7600 (10^{-23})$	$84.315 (10^{+23})$	64.079	268.108	1.810
650	377	$0.7600 (10^{-23})$	84.315 (10 <sup>+23</sup> )	64.079	268.108	1.810

MW (FOX-7) = 148.086 Grams/Mole

Table 3. FOX-7 Computed Specific Heat Results for FOX-7 From Reference 8, (Fig. 6)

Т	Т	C <sub>P</sub> * (Fig.6)	$C_{P}$	$C_{P}$	$C_P^{**}$
°K	°C	Joules  Cell Mole – °K	$\frac{\text{Joules}}{\text{Mole} - {}^{\circ}\text{K}}$	$\frac{\text{Cal.}}{\text{Mole} - {}^{\circ}\text{K}}$	Cal. Atom−°K
0	-273	0.00	0.00	0.0000	$0.0000 (10^{-23})$
50	-223	180.00	45.00	10.7553	0.1276 (10 <sup>-23</sup> )
100	-173	300.00	75.00	17.9254	0.2126 (10 <sup>-23</sup> )
150	-123	380.00	95.00	22.7055	0.2693 (10 <sup>-23</sup> )
200	-73	475.00	118.75	28.3819	0.3366 (10 <sup>-23</sup> )
250	-23	550.00	137.50	32.8633	0.3898 (10 <sup>-23</sup> )
300 (RT)	27	640.00	160.00	38.2409	0.4535 (10 <sup>-23</sup> )
350	77	705.00	176.25	42.1248	0.4996 (10 <sup>-23</sup> )
400 (EOD)	127	770.00	192.50	46.0086	0.5457 (10 <sup>-23</sup> )

<sup>\*</sup> The C<sub>P</sub> from Fig. 6 of Reference [8] is for 4 moles of FOX-7.

FOX-7 
$$\rightarrow$$
 C<sub>2</sub> H<sub>4</sub> N<sub>4</sub> O<sub>4</sub>  $\rightarrow$  q = 14.0 Atoms/Molecules

$$N_{AV}$$
 = Avogadro's number = 6.02252 (10<sup>+23</sup>) = Molecules/Mole

$$NAPMW = q N_{AV} = 84.315 (10^{+23}) = Atoms/Mole$$

\*\* 
$$C_P [Cal/(Atom - {}^{\circ}K)] = C_P [Cal/(Mole - {}^{\circ}K)] / (NAPMW)$$

RT = Room Temperature

EOD = End of Data

Table 4. FOX-7  $C_P(T)$  Prediction Comparison

Т	Т	$\mathbf{C}_{P_1}$	$\mathbb{C}_{\mathbb{P}_2}$	$\Delta C_{P}$	$\frac{\Delta C_P}{C_{P_1}} (100)$
°K	°C	$\frac{\text{Joules}}{\text{Mole} - {}^{\circ}\text{K}}$	Joules Mole – °K	Joules Mole – °K	Percentage Difference
		Ref. [8], (Table 3)	Present Report (Table 2)	C <sub>P2</sub> - C <sub>P1</sub>	%
0	.273	0.000	0.000	0.000	0.00
100	-173	75.000	70.555	-4.445	-5.93
150	-123	95.000	92.109	-2.891	-3.04
200	-73	118.750	113.664	-5.086	-4.28
250	-23	137.500	135.218	-2.282	-1.66
300 (RT)	27	160.00	156.773	-3.227	-2.02
350	77	176.250	178.363	+2.113	+1.20
400 (EOD)	127	192.500	199.917	+7.417	+3.85

$$\frac{\Delta C_P}{C_{P_1}}\big(100\big) = \left(\frac{C_{P_2} - C_{P_1}}{C_{P_1}}\right) \text{ x } 100.00 = \text{Percent Difference}$$

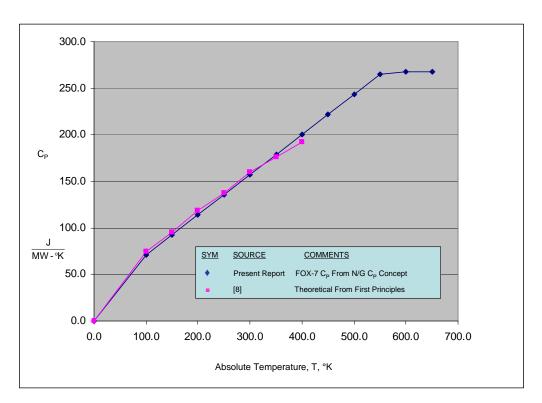


Figure 1. FOX-7 C<sub>P</sub> Prediction Comparison

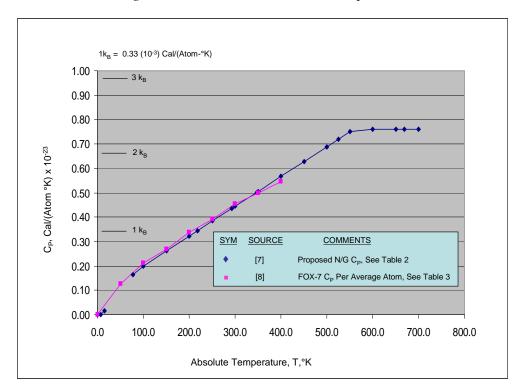


Figure 2. Comparison of the Proposed N/C  $C_P$  and the Calculated  $C_P$  for FOX-7 From Reference 8

#### REFERENCES

- 1. Billingsley, J. P., "Energetic Materials Shock Sensitivity Relevance to Specific Heat Properties," Technical Report RD-SS-95-2, July 1995, U.S. Army Missile Command, Redstone Arsenal, Alabama 35898.
- 2. Billingsley, J.P., Energetic Materials Shock Sensitivity Relevance to Specific Heat Properties," paper in Shock Compression of Condensed Matter 1995, AIP Conference Proceedings, 370, Part 1, Pages 429-432.
- 3. Billingsley, J.P., "Two Additional Examples of Energetic Materials Shock Sensitivity Correlations with Specific Heat Characteristics," Technical Report RD-SS-99-8, June 1999, U.S. Army Missile Command, Redstone Arsenal, Alabama, 35898.
- 4. Billingsley, J.P., "HMX and HNS Shock Sensitivity Correlation with Specific Heat and Reactive Temperature Magnitudes," Paper in Shock Compression of Condensed Matter 1999, AIP Conference Proceedings 505, Part II, Pages 899-902.
- 5. Billingsley, J.P., "Impact Shock Sensitivity of a TATB Based Explosive Relevant to Specific Heat Properties," Technical Report TR-AMR-SS-06-09, February, 2006, U.S. Army RDECOM, Redstone Arsenal, Alabama 35892.
- 6. Billingsley, J.P., "PBX-9502 Shock Sensitivity Correlation with Specific Heat and Reactive Temperature Magnitudes," paper presented at the AIP Shock Compression of Condensed Matter -2005 Conference, July 31 August 5, 2005.
- 7. Billingsley, J.P., "The Nominal/Generic Specific Heat Per Average Atom Concept for CHNO Energetic Materials," Technical Report AMR-SS-06-35, July 2006, U.S.Army RDECOM, Redstone Arsenal, Alabama 35892.
- 8. Zerilli, F.J., and Kuklja, M. M., "Equation of state of 1, 1-Diamino-2, 2-Dinitroethylene from First Principles," paper in Shock Compression of Condensed Matter -2005 Proceeding, AIP Conference Proceedings 845, 2006, pages 183-186, edited by M.D. Furnish: M. Elbert: T. P. Russell, and C.T. White.
- 9. Zerilli, F.J.,,and Kuklja, M.M., "Ab-Initio O°K Isotherm for organic Molecular Crystals," paper in Shock Compression of Condensed Matter 2003 Proceedings, AIP Conference Proceedings 706, 2004, pages 123-126, edited my M.D. Furnish, Y. M. Gupta, and J. W. Forbes.
- Kuklja, M.M., Rashkeev, S.N., and Zerilli, F.J., "Ab-Initio Calculations of the Electronic Structure of 1, 1-Diamino-2, 2-Dinitro-Ethylene," paper in Shock Compression of Condensed Matter – 2003 Proceedings, AIP Conference Proceedings 706, 2004, pages 363-366, edited by M.D. Furnish, Y. M. Gupta, and J. W. Forbes.
- 11. Ostmark, J., Langlet, A., Bergman, J., Wingborg, N., Wellmar, U., and Bemm, U., "FOX-7 A New Explosive with Low Sensitivity and High Performance," paper in Eleventh International Detonation Symposium Proceedings, Published as ONR 33300-5, 2000, pages 807-812.
- 12. Satoh, Shun-Ichi, "Heat Capacity and Chemical Constitution," Journal of Scientific Research Institute (Tokyo), Vol. 43, No.1194, 1948, pages 19-30.
- 13. Van Krevelen, D.W., Properties of Polymers, Third Completely Revised Edition, Elsevier Scientific Publishing Company, Amsterdam, 1990.
- 14. Van Krevelen, D.W., and Hoftyzer, P.J., "Properties of Polymers", Elsevier Scientific Publishing Company, Amsterdam, 1976.

# $\label{eq:appendix} \begin{array}{c} APPENDIX\ A\\ COMPUTATION\ OF\ m_{AV},\ d_{1av}\ and\ MW\ for\ FOX-7 \end{array}$

#### APPENDIX A

#### Computation of m<sub>AV</sub>, d<sub>1av</sub> and MW for FOX-7

The solid materials considered in this study were chemical mixtures. For these mixtures, the weighted average mass,  $m_{AV}$ , of a single atom in the material was desired.

First, it was necessary to compute the mass of a single atom for each of the elements contained in the solid. The solid was composed of one or more of the following elements:

Carbon, C; Hydrogen, H; Nitrogen, N; Oxygen, O:

The mass of a single atom of these elements is:

$$m = \frac{AW}{N_{AV}} = \frac{gram/(gram - mole)}{atoms/(gram - mole)} = \frac{grams}{atom}$$
 [A-1]

Where:

$$AW = \frac{grams}{gram - mole}$$

Nav = Avagadros Number = 
$$6.02252 \times 10^{+23} \frac{\text{atoms}}{\text{gram - mole}}$$

Table A-1 lists AW and m for each of the elements in the above list. Values of Nav and AW are from various chemistry text books and handbooks.

To compute the average weight  $(m_{AV})$  of an atom in the material, the chemical formula or proportional chemical composition is required. Of course, the weight (m) of each elemental atom must be known, since  $m_{AV}$  is just a weighted average of the elemental atoms in the material. The procedure is valid for mixtures of compounds as well as compounds. See References [1 & 3] for examples.

When  $m_{AV}$  is computed, then the average space between the atoms  $(d_{1AV})$  is given by the following relation:

$$d_{1av} = \left(\frac{m_{AV}}{\rho_{o}}\right)^{1/3} = cm$$
 [A-2]

Computations of  $m_{AV}$  and  $d_{1AV}$  for FOX-7 are included in this appendix. The Molecular weight, MW, is also calculated.

## Computation of $m_{AV}$ for FOX-7

Table A-1. Mass of a Single Atom for Selected Elements

Element	AW	$N_{AV}$	m
	<u>Grams</u>	<u>Atoms</u>	<u>Grams</u>
	Gram - Mole	Gram - Mole	Atom
Carbon (C)	12.011	$6.02252(10^{23})$	1.9943(10 <sup>-23</sup> )
Hydrogen (H)	1.008	$6.02252(10^{23})$	$0.1674(10^{-23})$
Nitrogen (N)	14.008	$6.02252(10^{23})$	2.3259(10 <sup>-23</sup> )
Oxygen (O)	16.00	$6.02252(10^{23})$	$2.6567(10^{-23})$

FOX-7 chemical composition: C<sub>2</sub> H<sub>4</sub> N<sub>4</sub> O<sub>4</sub> [8, 11]

Table A-2. FOX-7  $m_{AV}$  Computation

$$C_2$$
 2 x 1.9943 (10<sup>-23</sup>) = 3.9886 (10<sup>-23</sup>) Grams  
 $H_4$  4 x 0.1674 (10<sup>-23</sup>) = 0.6696 (10<sup>-23</sup>) Grams  
 $N_4$  4 x 2.3259 (10<sup>-23</sup>) = 9.3036 (10<sup>-23</sup>) Grams  
 $O_4$  4 x 2.6567 (10<sup>-23</sup>) = 10.6268 (10<sup>-23</sup>) Grams  
14 Atoms weigh 24.5886 (10<sup>-23</sup>) Grams

$$m_{AV} = \frac{24.5886 (10^{-23}) \, Grams}{14 \, Atoms} = 1.756329 (10^{-23}) \frac{Grams}{Atoms}$$
 $\approx 1.7563 (10^{-23}) \, Grams \, per \, average \, Atom$ 

#### Computation of $d_{1AV}$ for FOX-7

Reference [11] gives the following density values for FOX-7.

 $\rho_{\circ} = 1.65 \text{ Grams/CC (for small scale gap test)}$ 

 $\rho_{\circ} = 1.878 \text{ Grams/CC (for crystal density)}$ 

For 
$$\rho_{\circ} = 1.65 \text{ Grams/CC}$$

$$d_{1AV}^{3} = \frac{m_{AV}}{\rho_{\circ}} = \frac{17.563 (10^{-24})}{1.65} \frac{Grams}{Grams/CC} = CM^{3}$$

$$= 10.644242 (10^{-24}) CM^{3}$$
 $d_{1AV} = 2.199741 (10^{-8}) CM = 2.1997 A$ 

For 
$$\rho_{\circ} = 1.878 \text{ Grams/CC}$$

$$d_{1AV}^{3} = \frac{m_{AV}}{\rho^{\circ}} = \frac{17.563(10^{-24})}{1.878} \frac{Grams}{Grams/CC} = CM^{3}$$

$$= 9.351970 (10^{-24}) CM^{3}$$
 $d_{1AV} = 2.106852 (10^{-8}) CM = 2.1069 A$ 

These values for  $d_{1AV}$  are much larger than C(1)-C(2) bond lengths  $(1.456\,A)$  and N(12)-O(22) bond lengths  $(1.242\,A)$  given in Reference [11]. However, Figure 6 of Reference [11] shows a layered structure of the molecules where the minimum separation distance between nitrogen and oxygen atoms is  $3.11\,A$ . So the average distance between atoms is approximately  $2.10\,A$  as shown in the above computations.

## **Computation of MW for FOX-7**

Chemical formula =  $C_2 H_4 N_4 O_4 =$   $C_i H_j N_k O_l$ 

Table A-3. MW Calculation for C<sub>2</sub> H<sub>4</sub> N<sub>4</sub> O<sub>4</sub>

Element	AW	n	n x AW	
	grams	~	grams	
	gram - mole		gram - mole	
Carbon (C)	12.011	i = 2	24.022	Grams
Hydrogen (H)	1.008	j = 4	4.032	Grams
Nitrogen (N)	14.008	k = 4	56.032	Grams
Oxygen (O)	16.000	1 = 4	64.000	Grams
M	$W = \sum_{n \in \mathbb{N}} n \times AW$	, =	148.086	grams mole

So MW = 148.086 Grams  $\approx 148$  Grams

NAPMW = 
$$\frac{MW}{m_{AV}}$$
 =  $\frac{148.086}{1.756329(10^{-23})} \frac{Grams}{\left(\frac{Grams}{Atom}\right)}$  = Atoms = 84.315638 10<sup>+23</sup> Atoms per mole

This checks well with:

NAPMW = 
$$q N_{AV}$$
 =  $(14.0) (6.02252 * 10^{+23})$   
=  $84.315280 * 10^{+23}$ 

Where: 
$$q = \sum n = i + j + k + l = 2 + 4 + 4 + 4 = 14 \frac{Atoms}{Molecule}$$

 $N_{AV} = Avogadro's number - No. of molecules per MW.$ (Molecules/MW)

# $\label{eq:appendix B} \textbf{APPENDIX B} \\ \textbf{SPECIFIC HEAT (C_P) UNITS CONVERSION RELATIONS} \\$

#### Appendix B

#### Specific Heat (C<sub>P</sub>) Conversion Relations

Experimental Specific heat data are usually given in calories (cal) or Joules (J) per molecular weight (MW) per ° centigrade (°C) or ° Kelvin (°K). That is:

$$C_P = C_P$$
 (cal. per MW per ° C or °K)

or

$$C_P = C_P$$
 (J per MW per °C or °K).

Occasionally, C<sub>P</sub> is presented as cal. or J per gram °C or °K.

$$C_P = C_P$$
 (cal per gram per °C or °K)

or

$$Cp = Cp (J per gram per °C or °K)$$

Note that:

Cp (cal. or J per gram per 
$$^{\circ}$$
C of  $^{\circ}$ K) = Cp (cal, or J per MW per  $^{\circ}$ C or  $^{\circ}$ K)/MW.

Since interest is in Cp (cal or J per atom per °C or °K) then, Cp (cal. or J per atom per °C or °K =  $m_{AV}$  Cp (cal. or J per gram per °C or °K).

See Appendix A for procedures to determine  $m_{AV}$ . Note that  $1^{\circ}$  C =  $1^{\circ}$  K, and that the conversion factor between Joules and calories is 4.184 so that:

Cp (J per atom per 
$$^{\circ}$$
K) = 4.184 \* Cp (cal per atom per  $^{\circ}$ K).

Also, since 1 Joule = 
$$10^7$$
 ergs =  $10^7$  grams (cm/sec)  $^2$ 

then,

Cp grams 
$$(cm^2/sec^2)$$
 per atom per °K =  $10^7$ . (J per atom per °K).

The above relations were employed in Reference 2, and the present report. However, Cp can also be given in terms of a velocity squared per °K as follows:

$$\begin{split} &Cp(cm^2/sec^2~per~^\circ K)=10^7~x~Cp~(J~per~gram~per~^\circ K).~In~these~units~for~Cp,~\Delta(v.e.)_{TR}~=\\ &\int_{T_{EXP}}^{T_R} C_p dT~=(cm/sec)^2,~so~U_{PCR1}~=~\sqrt{\Delta(v.e.)_{TR}}~=and~U_{PCR2}~=~\sqrt{2\Delta(v.e)_{TR}}~=~\sqrt{2}~U_{PCR1}~. \end{split}$$

This eliminates having to compute an average mass  $(m_{\rm AV})$  and the  $C_P$  in calories or Joules per average atom.

# INITIAL DISTRIBUTION LIST

		<u>Copies</u>
Weapon Systems Technology Info ATTN: Ms. Vakare Valaitis 1901 N. Beauregard Street, Suite 4 Alexandria, VA 22311-1720	rmation Analysis Center (WSTIAC)	1
Defense Technical Information Ce 8725 John J. Kingman Rd., Suite O Fort Belvoir, VA 22060-6218		1
AMSRD-AMR		Electronic
AMSRD-AMR-IN-IC		2
AMSRD-L-G-I,	Anne Lanteigne	1
Los Alamos National Laboratory ATTN: Dr Steven A. Sheffield/Dr. Group DX-1 Mail Stop P-952 Los Alamos, NM 87545	R.L. Gustavsen	1
Los Alamos National Laboratory ATTN: Dr. Phillip M. Howe Mail Stop 945 Los Alamos, NM 87545		1
Los Alamos National Laboratory ATTN: Roberta Mulford Mail Stop NMT-15 Los Alamos, NM 87545		1
Los Alamos National Laboratory ATTN: Davis Tonks Dean Preston X-7 Mail Stop F699 Los Alamos, NM 87545		1
Los Alamos National Laboratory ATTN: Dr. Douglas G. Tasker Mail Stop J566 Los Alamos, NM 87545		1

Los Alamos National Laboratory	Copies 1
ATTN: George Gray III Mail Stop G-755 Los Alamos, NM 87545	1
Los Alamos National Laboratory ATTN: James N. Johnson/James D. Johnson Mail Stop B-221 Los Alamos, NM 87545	1
Los Alamos National Laboratory ATTN: Blain Asay Mail Stop C-920 Los Alamos, NM 87545	1
Los Alamos National Laboratory ATTN: Joseph N. Fritz DX-1 Mail Stop P-953 Los Alamos, NM 87545	1
Los Alamos National Laboratory ATTN: Dr. Joseph C. Foster, Jr./Dr. Y. Horie 34 Parr Circle Ft. Walton Beach, FL 32548	1
Sandia National Laboratory ATTN: Dr. Mark B. Boslough Shock Wave and Explosion Physics P.O. Box 5800, Division 1153 Albuquerque, NM 87185	1
Dr. Dennis Hayes Consultant Box 591 Tijeras, NM 87059	1
Sandia National Laboratory ATTN: Dr. M. D. Furnish/Dr. L.C. Chhabildas Mail Stop 1181 P.O. Box 5800 Albuquerque, NM 87185-1191	1

Sandia National Laboratory	Copies 1
ATTN: Marlin E. Kipp Mail Stop 0820 P.O. Box 5800 Albuquerque, NM 87185-0820	
Mr. Jon Maienschein/Dr. L. Fried/Dr. F. Garcia Lawrence Livermore National Laboratory Mail Code L-282 P.O. Box 808 Livermore, CA 94551	1
Dr. Craig M. Tarver/Dr. P.A. Urtiew/Dr. K.S. Vandersall Lawrence Livermore National Laboratory Mail Code L-282 P.O. Box 808 Livermore, CA 94551	1
Dr. Neil Holmes Lawrence Livermore National Laboratory Mail Code L-041 P.O. Box 808 Livermore, CA 94551	1
U.S. Army Picatinny Arsenal Energetic Materials Division, Bldg. 3028 ATTN: Dr. Surya N. Bulusu Picatinny, NJ 07806-5000	1
Naval Surface Warfare Center – Indian Head ATTN: Dr. Ruth Doherty 101 Strauss Avenue Indian Head, MD 20640-5035	1
Naval Surface Warfare Center – Dahlgren Division ATTN: Mr. Daniel Vavrick Code G-22 17320 Dahlgren Road Dahlgren, VA 22448-5100	1
Naval Surface Warfare Center – Indian Head ATTN: Dr. C.S. Coffey/ Dr. J. M. Short 101 Strauss Avenue Indian Head, MD 20640-5035	1

Naval Research Laboratory	Copies 1
Dynamics of Solids Branch Condensed Material and Radiation Sciences Division ATTN: Mr. Andrew E. Williams Washington, DC 20375	•
U. S. Army Research Laboratory ATTN: Mr. Douglas E. Kooker AMSRL-WM-TB Aberdeen Proving Ground, MD 21005-5066	1
U.S. Army Research Laboratory 1 ATTN: Dr. Dattatraya Dandekar Aberdeen Proving Ground, MD 21005-5066	
Director U.S. Army Research Laboratory SLCRO-MS ATTN: Dr. Kailasam Iyer/Dr. Michael Cistan Research Triangle Park, NC 27709-2211	1
U.S. Army Research Office ATTN: Mr. David Mann P.O. Box 12211 Research Triangle Park, NC 27709-2211	1
Eglin Air Force Base ATTN: (AFMC) Dr. Bill Dyess 46 Test Wing Eglin AFB, FL 32542-5910	1
Wright-Patterson Air Force Base ATTN: Dr. Ted Nicholas WRDC-MLLN Wright-Patterson, AFB, OH 45433	1
Director U.S. Army Research Laboratory AMSRD-ARL-WM-TD ATTN: Dr. S.B. Segletes Aberdeen Proving Ground, MD 21995-5066	1
Director U.S. Army Research Laboratory AMSRD-ARL-SL-BE ATTN: Dr. E.G. Davis Aberdeen Proving Ground, MD 21995-5066	1

California Institute of Technology Division of Geological and Planetary Sciences ATTN: Dr. Thomas J. Ahrens MS-252-21 Pasadena, CA 91125	<u>Copie</u> 1
University of Texas at Austin Department of Aerospace Engineering and Engineering Mechanics ATTN: Dr. Byron Tapley W.R. Woolrich Labs Austin, TX 78712-1085	1
University of Dayton Research Institute Impact Physics Group ATTN: Dr. N. S. Brar 300 College Park Avenue Dayton, OH 45469-0182	1
U.S. Naval Academy ATTN: Dr. Mark Ehlert Chemistry Department Annapolis, MD 21402	1
University of California – Los Angeles ATTN: Professor John J. Gilman 6532 Boelter Hall Los Angeles, CA 90024	1
Argonne National Laboratory Technical Information Services Report Unit Bldg. 203 Argonne, IL 60439	1
Dr. E.R. Fitzgerald Box 291 Monkton, MD 21111	1
Institute for Advanced Technology 1 ATTN: Dr. Stephen Bless/Dr. Harry Fair 4030 Braker Lane, Ste. 200 Austin, TX 78759	

	<u>Copies</u>
University of Denver Denver Reasearch Institute ATTN: Mr. Larry Brown Denver, CO 80108	1
Washington State University Institute for Shock Physics Department of Physics ATTN: Dr. Y. M. Gupta/Dr. James R. Asay Pullman, WA 99164-2816	1
Center for Explosive Technology Research Campus Station ATTN: Dr. Pers-Anders Person Socorro, NM 87801	1
Southwest Research Institute Engineering Dynamics Department ATTN: Dr. Charles E. Anderson, Jr./Dr. James Walker P.O. Drawer 28510 San Antonio, TX 78228-0510	1
University of Delaware Department of Chemistry and Biochemistry ATTN: Professor Thomas B. Brill Newark, DE 19716	1
Georgia Institute of Technology School of Materials Science and Engineering ATTN: Dr. N.N. Thadhani 771 Ferst Drive Atlanta, GA 30332-04245	1
Brown Unversity Division of Engineering ATTN: Dr. R.J. Clifton 182 Hope Street Providence, RI 02912	1
Lynn Barker Valyn International Box 11647 Albuquerque. NM 87192	1

	<u>Copies</u>
Dr. Leonard. I. Stiel Polytechnic University 6 Metrotech Center Brooklyn, NY 11201	1
Zernow Technical Services, Inc. ATTN: Dr. Louis Zernow 425 West Bonita Avenue, Suite 208 San Dimas, CA 91773	1
Alan M. Frank and Associates ATTN: Dr. Alan M. Frank 748 Wimbledon Lane Livermore, CA 94559	1
SRI International ATTN: Dr. Paul DeCarli/Dr. L. Seaman 333 Ravenswood Avenue Menlo Park, CA 94025	1
Dynetics, Inc. ATTN: Mr. James Miller 1000 Explorer Blvd. Huntsville, AL 35806	1
Davidson Technologies, Inc. ATTN: Mr. Shane Strickland Mr. Chris Pitts Ms. Judy Smith 530 Discovery Drive Huntsville, AL 35806	1 1 1
Morgan Research Corporation ATTN: Mr. Brent Deerman/Mr. Scott Hill 4811 A Bradford Drive Huntsville, AL 35805-1948	1
Enig Associates, Inc. ATTN: Mr. Julius W. Enig 12501 Prosperity Drive # 340 Silver Spring, MD 20404	1
International Research Associates ATTN: Dr. Dennis L. Orphal 4450 Black Avenue, Suite E Pleasanton, CA 94566	1

	Copies
Vitali F. Nestorenko UCSD 9500 Gilman Drive	1
La Jolla, CA 92093-0411	
Dr. Julius Roth Consultant 308 Canyon Drive Portola, CA 94025	1
Dr. Franklin E. Walker Interplay 584 W. Treeline Dr Alpine, UT 84994	1
Dr. Muril Robertson Gleason Research Associates 2227 Drake Avenue SW Suite 2 Huntsville, AL 35805	1
Mr. Hugh R. James AWE, Aldermaston Bldg. E1 RG7 4PR, Reading, Berkshire, UK	1
Mr. Frederic Peugeot Technical Specialist MSIAC, NATO Headquarters B-1110, Brussels Belgium	1
Dr. Manfred Held TDW/EADS 86532 Schroben Hausen, Germany	1
Dr. Dana D. Dlott University of Illinois 600 S. Matthews Avenue Urbana, IL 61801	1
Alliant, Inc. ATTN: A. Garn Butcher P.O. Box 98 Magna, UT 84037	1

Dr. Joe Carleone Sr. VP and Chief Products Officer Irvine Sensors Corporation 3001 Red Hill Ave., Bldg 4, Suite 108 Costa Mesa, CA 962626-453	Copies 1
Applied Research Associates, Inc. ATTN: Dr. Dennis D. Grady 4300 San Mateo Blvd., NE: Suite A-220 Albuquerque, NM 87110	1
ITT Industries Inc. ATTN: Dr. James Wilbeck 6767 Old Madison Pike Huntsville, AL 35806	1
Raytheon Company ATTN: W. Zar 1151 E. Hermans Rd Bldg. 810, M/S 8 Tucson, AZ 85706	1
Brigs Company ATTN: Joseph E. Backofen 4192 Hales Ford Road Moneta, VA 24121	1
Dr. S.K. (Jim) Chan 4076 Kuchiran St. Pierrefonds, P Q Canada, H9H4A9	1
Dr. Martin Braithwaite Imperial College London Department of Chemical Engineering and Chemical Technology Room 336, Roderic Hill Building South Kensington Campus London, SW7 2AZ, England	1

U.S. Army Space and Missile Def Mr. Mike Schexnayder Deputy Director USASMDC/ARSTRAT Bldg 5220 Redstone Arsenal, AL 35898	ense Command	Copies 1
U.S. Army Space and Missile Def Mr. Kirk Newman SMDC-RD-ICM Bldg 5220 Redstone Arsenal, AL 35898	ense Command	1
Dr. F.J. Zerilli Naval Surface Warfare Center Research and Technology Departr Indian Head, MD 20640	ment	1
Dr. M.M. Kuklja National Science Foundation Division of Materials Research Arlington, VA 22230		1
AMSRD-AMR-AS-TI	Mr. Tommy Harris Mr. Frank Wlodarski	1 1
AMSRD-AMR-PS	Mr. Steve Cornelius Dr. Mike Lyon	1 1
AMSRD-AMR-PS-PR	Mr. W.M. Chew Mr. Jay Lilley Mr. R.W. Milton Mr. D.M. Thompson	1 1 1 1
AMSRD-AMR-PS-WF	Mr. Jason Gilliam Mr. Scott Howard Mr. Greg Johnson Dr. Darin Kielsmeier Mr. Donald Lovelace Mr. Adolphus McDonald Mr. Allen Stults Mr. Paul Turner	1 1 1 1 1 1 1

		Copies
AMSRD-AMR-SG-RF	Dr. Brian Smith	1
AMSRD-AMR-SS	Mr. Greg Tackett	1
AMSRD-AMR-SS-AT	Mr. Richard Kretzschmar	1
AMSRD-AMR-SS-AV	Mr. Steve Low	1
AMSRD-AMR-SS-EG	Dr. J.P. Billingsley Mr. Jason Brister Mr. Dustin Clark Ms. Edith Crow Mr. Brian Harrison Ms. Ann Kissell Ms. Susan Parker Mr. Allen Pike Mr. Glenn Romanczuk Mr. William Schrenk Mr. Dan Shady Ms. Kimberly Williams	10 (paper) 1 (electronic)  1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
AMSRD-AMR-SS- HW	Mr. Alex Jolly	1
AMSRD-AMR-SS-MD	Mr. James Grabney/Mr. Brad G	ass 1
AMSRD-AMR-SS-TM	Mr. Scott Speigle	1
AMSRD-AMR-WS	Dr. J. S. Bennett/Dr. Holloman	1
SFAE-MSL-CWS-E	Mr. Al Dykstra	1
SFAE-MSLS-JAMS	Mr. Chuck Allen/ Mr. Ralph Par	rker 1